UNCLASSIFIED

UNCLASSIFIED			
AD NUMBER			
AD448060			
LIMITATION CHANGES			
TO:			
Approved for public release; distribution is unlimited.			
FROM: Distribution authorized to U.S. Gov't. agencies and their contractors; Administrative/Operational Use; AUG 1964. Other requests shall be referred to Army Missile Command, Redstone Arsenal, AL.			
AUTHORITY			
dod, 2 aug 1966			

UNCLASSIFIED

AD_448060

DEFENSE DOCUMENTATION CENTER

FOR

SCIENTIFIC AND TECHNICAL INFORMATION

CAMERON STATION ALEXANDRIA. VIRGINIA



UNCLASSIFIED

NOTICE: When government or other drawings, specifications or other data are used for any purpose other than in connection with a definitely related government procurement operation, the U. S. Government thereby incurs no responsibility, nor any obligation whatsoever; and the fact that the Government may have formulated, furnished, or in any way supplied the said drawings, specifications, or other data is not to be regarded by implication or otherwise as in any manner licensing the holder or any other person or corporation, or conveying any rights or permission to manufacture, use or sell any patented invention that may in any way be related thereto.

448060

CONTRACT TECHNICAL NOTE

SOME LIMITATIONS ON THE PERFORMANCE
OF HIGH-VELOCITY GUNS

4 4 8 0 6 0

CONTRACT NO. DA-04-495-ORD-3567(Z)

HYPERVELOCITY RANGE RESEARCH PROGRAM

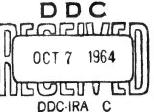
A PART OF PROJECT "DEFENDER"

GM DEFENSE RESEARCH LABORATORIES

SANTA BARBARA, CALIFORNIA



AEROSPACE OPERATIONS DEPARTMENT





CTN64-03

AUGUST 1964

NOTICE

The information presented in this Technical Note is preliminary and should not be interpreted as representing the final opinion of GM Defense Research Laboratories or the author on the subject. Further, to expedite prompt distribution of the technical contents, no attempt at an editorial review has been made.

ARPA ORDER NO. 357-63

Copy No 111

PROJECT CODE NO. 7400

CONTRACT TECHNICAL NOTE

SOME LIMITATIONS ON THE PERFORMANCE OF HIGH-VELOCITY GUNS

John S. Curtis

THIS RESEARCH WAS SUPPORTED BY THE ADVANCED RESEARCH PROJECTS AGENCY AND WAS MONITORED BY THE U.S. ARMY MISSILE COMMAND REDSTONE ARSENAL, ALABAMA

GM DEFENSE RESEARCH LABORATORIES

SANTA BARBARA, CALIFORNIA



AEROSPACE OPERATIONS DEPARTMENT

CONTRACT NO. DA-04-495-ORD-3567(Z)

HYPERVELOCITY RANGE RESEARCH PROGRAM

A PART OF PROJECT "DEFENDER"

CTN64-03

Qualified requesters may obtain copies of this report from DDC

AUGUST 1964

SOME LIMITATIONS ON THE PERFORMANCE OF HIGH-VELOCITY GUNS

Free-flight ranges using light-gas gun launchers are an important tool in the investigation of the phenomena associated with hypersonic flight. The successful completion of space-flight projects currently planned requires that the velocity capability of research facilities be extended to reentry and even meteoric speeds. If the planning of research using free-flight ranges is to be done intelligently, it is necessary to know what the ultimate capability of light-gas gun launchers will be. However, no clearly defined theoretical upper limit to the velocity capability of a light-gas gun exists. A moderately well defined upper limit does exist for the size of projectiles which can be launched at any specified velocity in any given gun.

It is the purpose of this note to discuss the velocity regime over which the light-gas gun will be useful as a research tool. This will be done by considering the ratio of accelerating forces to the strength of the projectile.

The velocity achieved in any launcher is the product of the acceleration and the time of launch if the acceleration is constant, and it is the integral under the acceleration-time curve for variable-acceleration launches. To achieve high velocity, either acceleration or time, or both, must be large. In a light-gas gun, the accelerating force is generated by the pressure of the light gas acting on the base of the projectile. Increasing the accelerating force implies increasing gas pressure, and utlimately a pressure limit is reached. If the time of launch is increased,

the length of the launch tube must be increased; the length of the launch tube is proportional to the square of the launch time for a given velocity. Thus, in order to limit the size of the gun, it is very desirable to make the acceleration, and hence the pressure, as high as possible.

Two factors operate to limit the pressure which can be used in a light-gas gun. First, and most important, the pressure applied to the base of the projectile must not be so large as to break or permanently deform the model. Second, the pressure in the gun should not overstress or destroy the gun. In general, the pressure which can be applied to a projectile of any sophistication without destroying it is an order of magnitude less than the pressure which can be withstood by a well designed gun.

In any given gun, the maximum projectile velocity will be achieved if the pressure applied to the base of the projectile is held constant at the maximum allowable value during the launching run. A gun which is able to operate in this manner is defined as an ideal gun (see Reference 1).

In an actual operating gun it is difficult, impractical, and in fact undesirable, to hold the base pressure exactly constant during the launching run. Since the ideal gun will impart the highest velocity to any given projectile, a measure of the quality of a gun can be made by comparing the performance of the actual and ideal gun at equal maximum base pressures. The ratio of the velocity obtainable in the ideal gun to the velocity obtained in the actual gun when the maximum base pressures are equal will be called the performance

factor. Thus, the performance factor for any launching sequence is the ratio of maximum base pressure to the average base pressure during the launch.

A gun has been built at GM DRL with the specific objective of improving the performance factor (see Reference 2). While no actual measure of the performance factor has been made, both computations and experience with the gun indicate that it operates consistently with a performance factor of two or less. By contrast, the performance factor of early light-gas guns is near five, and that of shock-heated light-gas guns probably lies between five and seven.

Next to pressure the main constraints are the limitations on gun performance which are imposed by the projectile strength. A graph giving the combinations of acceleration and distance which combine to yield various velocities is shown in Figure 1. It will be noted that for a distance of the order of 20 feet, an acceleration of the order of one million g's is required to achieve a velocity in the region between 30,000 and 40,000 feet per second.

The compressive stresses produced in a body by an acceleration of one million g's are shown in Figure 2. The body, for simplicity, is assumed to be a right cylinder, unsupported on the sides. The accelerating force is applied to the aft end of the cylinder and is distributed uniformly across the face of the cylinder. A constant stress gradient will exist in the body and the compressive stress will vary from zero at the front face to some maximum value at the rear face. The stress at any point will depend on the density of the material, the acceleration, and the distance from the front face of the material. The slope of the curves in Figure 2 is a function of density only, and

the curves have been computed for the specific materials which are so labeled. It will be noted that when accelerated in the manner specified at one million g's, the compressive stress one inch from the front face will be larger than the ultimate strength of most materials. Figure 3 is a generalization of Figure 2, wherein the stresses are shown as a function of length for a range of accelerations from 100,000 g's to 10,000,000 g's.

The usefulness of these simple curves in designing a gun for a particular experiment can best be shown by an example. Suppose we wish to launch an aero-dynamic model configuration at a velocity of 40,000 feet per second. The projectile configuration can be approximated by a cylinder one inch long. From a consideration of strength-to-weight ratio and the total projectile weight, we construct the projectile of aluminum having an ultimate compressive strength of 60,000 psi. What is the minimum size gun which will successfully launch this model?

First, enter the top part of Figure 3 at a compressive stress of 60,000 psi. Move horizontally to the aluminum line, then vertically downward from this point to the curved line labeled one inch. Read on the ordinate at this point an acceleration of 620,000 g's. This is the maximum acceleration which the projectile can withstand without exceeding the ultimate strength of the material. Before entering Figure 1 to determine the size of gun necessary, we must know the average acceleration. From the definition of the performance factor, the average acceleration is equal to the maximum acceleration divided by the performance factor. Using a performance factor of two as a representative

value, we get an average acceleration of 310,000 g's. Now entering Figure 1 at an acceleration of 310,000 g's, move vertically upward to the 40,000 feet per second line and at this point read the distance required on the ordinate as 82 feet. This is then the length of the barrel of the gun required.

With one major dimension of the gun fixed, it is possible to determine the remaining dimensions of the gun within reasonable limits. A barrel which has a length of 300 calibers is a long barrel in which to maintain a performance factor of 2. It is not unreasonable, however, to expect that such performance can be achieved. Using a length of 300 calibers, the diameter is then approximately 3. 3 inches. For an accelerated-reservoir gun, the diameter of the pump tube should be at least four times the diameter of the launch tube; so the pump tube will have a diameter of about one foot. The length of the pump tube will be determined by the compression ratio necessary to achieve a performance factor of two.

Using the method described in the preceding example, Figure 4 has been prepared showing the length of gun barrel required to launch aluminum cylinders of varous lengths at various velocities. From this figure it is apparent that it will require a very large gun to launch an aluminum projectile whose equivalent length is one inch at a velocity near 40,000 feet per second. (Equivalent length is defined as the length of a cylinder of the same material and weight in which the maximum compressive stress will be the same as that in the projectile when fired under the same conditions.)

The curves are also useful in determining the maximum velocity at which a projectile of a given length can be launched in a specified gun. Figure 3 and the performance factor are used as before, but Figure 2 is entered using the barrel length as the distance and the acceleration obtained from Figure 3.

A rough check on the validity of this approach has been obtained in the Ballistics Range at GM Defense Research Laboratories. Two projectiles have been fired a sufficient number of times to determine the approximate upper limit to the velocity with which they can be launched intact using a .22-caliber launch tube four feet long. The projectiles are a 3/16-inch-diameter solid aluminum sphere and a 1/8-inch-diameter solid glass sphere. Using 60,000 psi and 50,000 psi, respectively, as the maximum allowable compressive strengths and an equivalent projectile length of 2/3 diameter, the maximum velocities obtained from Figures 2 and 3 are 25,000 feet per second and 29,000 feet per second, respectively. The highest velocity at which a successful launch has been made using these projectiles is 22,000 and 26,700 feet per second, respectively. The agreement is quite satisfactory considering the very simple approach taken.

This analysis leads to three direct conclusions:

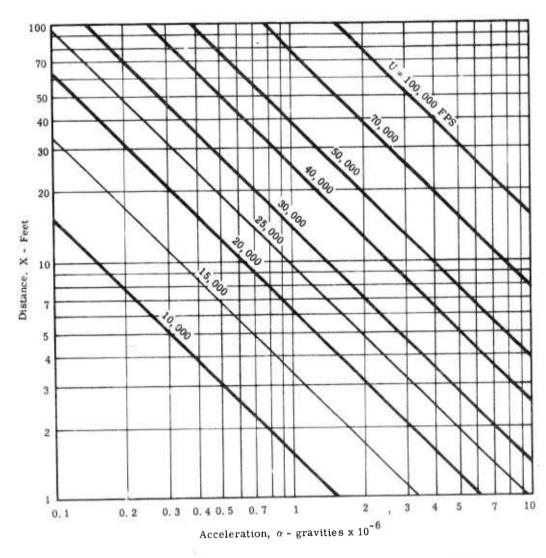
1. If the size and properties of the projectile are constant, the minimum linear dimensions of a gun necessary to launch the projectile are proportional to the square of the launch velocity, and the mass of the gun or any gun part is proportional to the sixth power of the launch velocity.

- 2. If the velocity of the launch is constant, the minimum linear dimensions of a gun necessary for successful launch are directly proportional to the length of the projectile, and the mass of the gun or any gun part is proportional to the cube of the length of the projectile.
- 3. If the size of the gun is constant, the length of the projectile which can be successfully launched is inversely proportional to the square of the launch velocity.

It is thus apparent that the size of gun necessary for any test is a critical function of the velocity of the test and the size of the projectile. Also, if free-flight range testing is to be done at a velocity of 40,000 feet per second, the size of the model must be small regardless of the size of the gun.

REFERENCES

- A. C. Charters and John S. Curtis, "High Velocity Guns for Free-Flight Ranges," Paper presented at Agard Specialists' Meeting, Belgium, Apr 1962; also published as GM Defense Research Laboratories Technical Memorandum TM62-207
- John S. Curtis, "An Accelerated-Reservoir Light-Gas Gun," NASA
 TN D-1144, 1962



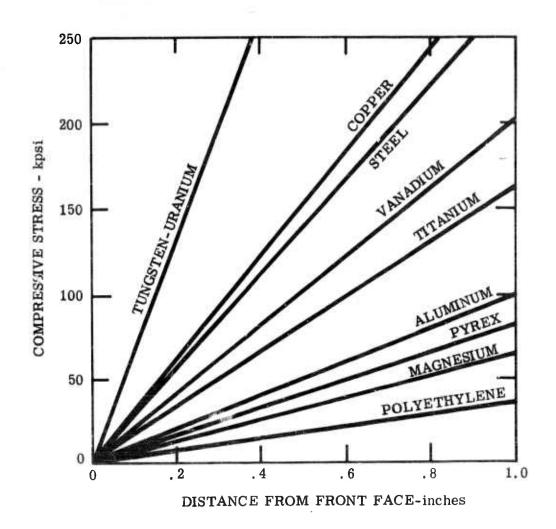


Figure 2 Compressive Stress as a Function of Distance From Front of a Cylindrical Projectile Accelerated by a Force Applied to One End, Resulting in an Acceleration of One Million Gravities

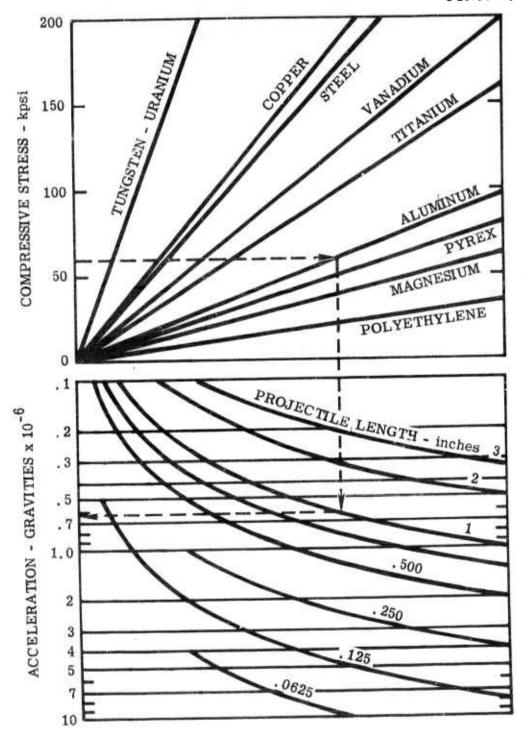


Figure 3 Nomograph

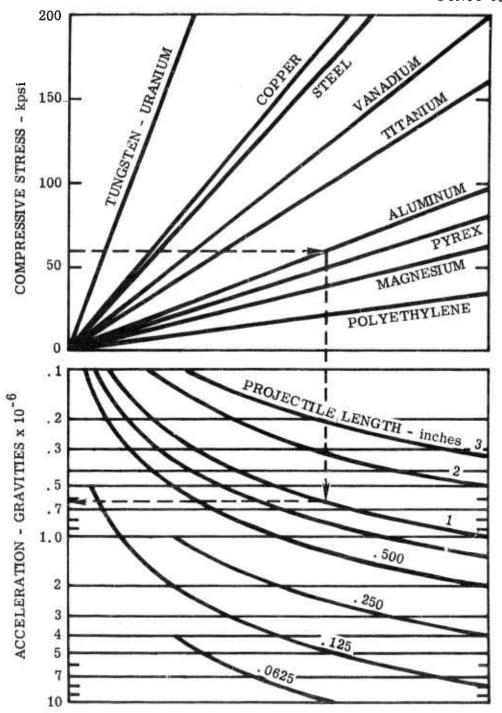


Figure 3 Nomograph

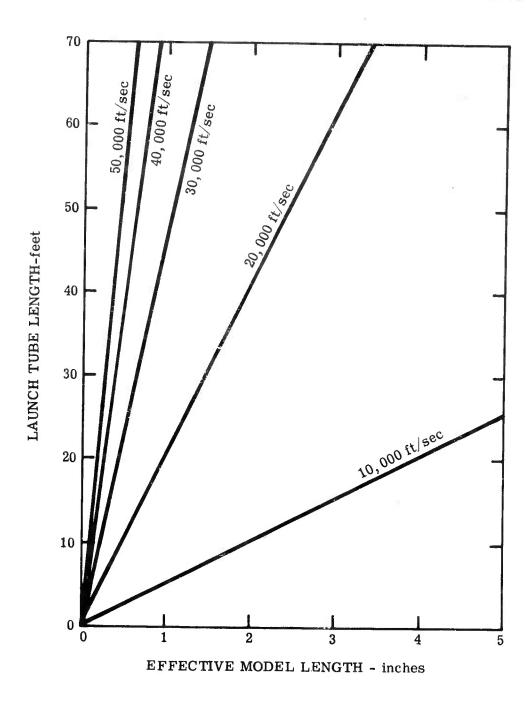


Figure 4 Launch Tube Length Required to Launch Aluminum Models of Various Effective Lengths

DISTRIBUTION LIST for Analysis Reports on the HYPERVELOCITY RANGE RESEARCH PROGRAM

Recipient	Copy No.	Recipient	Сору №.
Director Advanced Research Projects Agency Washington 25, D. C. ATTN: F. Koether ATTN: E. Haynes ATTN: C. McLain	1 - 3 4 5	U.S. Air Force Ballistic Systems Division AF Unit Post Office Los Angeles 45, California ATTN: Major W. Levy ATTN: Lt. K.G.Jefferson	51 52
ATTN: Major J. Kiernan Aerojet-General Corporation P. O. Box 296 Azusa, California	6	HQ BSD (AFSC) AF Unit Post Office Los Angeles 45, California ATTN: BSRVD	53
ATTN: Technical Library Aeronutronics Division, Ford Motor Co. Ford Road Newport Beach, California	7	USAF Cambridge Research Laboratorie Laurence Hanscom Field Bedford, Massachusetts ATTN: CRRELR, Stop 29	es 54
ATTN: Technical Information Services ATTN: R. Hoglund Aerospace Corporation 2400 E. El Segundo Blvd	8 9	Director USAF Office of Scientific Research Washington 25, D. C. ATTN: Mechanics Division/ Major Terrell	55
El Segundo, California ATTN: Manager of Penetration Aids Aerospace Corporation P. O. Box 95085 Los Angeles 45, California	10	Director Ames Research Center Moffett Field, California ATTN: H. Allen	56
ATTN: J. Logan Aerospace Corporation San Bernardino, California ATTN: Mr. R. Fowler ATTN: Mr. Howard Meyers	11 12 13-42	Applied Physics Laboratory The John Hopkins University 8621 Georgia Avenue Silver Spring, Maryland ATTN: G. Seielstad	57
Avco-Everett Research Laboratory 2385 Revere Beach Parkway Everett 49, Massachusetts ATTN: Dr. Bennett Kivel	43	Applied Physics Laboratory Sylvania Elec. Products Waltham, Massachusetts ATTN: R. Row	58
ATTN: Technical Library Avco Research and Advanced Developme Wilmington, Massachusetts ATTN: Dr. A. Pallone	45	Armour Research Foundation 10 W. 35th Street Chicago 16, Illinois ATTN: Fluid Dynamics Research Division	59
ATTN: Dr. W. E. Gibson ATTN: Dr. J. Ekerman Bell Telephone Laboratories Murray Hill, New Jersey ATTN: Dr. I. Pelech ATTN: Dr. S. P. Morgan ATTN: Dr. S. J. Buchsbaum	46 47 48 49 50	Commanding General U. S. Army Air Defense Command Colorado Springs, Colorado ATTN: Advanced Projects Division, G-	3 60

DISTRIBUTION LIST (continued)

Recipient	Copy No.	Recipient	Copy No.
Commanding General U.S. Army Ballistics Research Laboratories		Avco Corporation Research and Advanced Development	
Aberdeen Proving Gound, Maryland		Division Wilmington Magazahugatta	
ATTN: C. H. Murphy	61	Wilmington, Massachusetts ATTN: Technical Library	76
ATTN: B. J. Karpov	62	ATTN: Dr. Wentink	77
Commanding General	63	Special Projects Office	
U.S. Army Elec. and Communications		Department of The Navy	
Command		Washington 25, D. C. ATTN: Martin Bloom/SP-272	78
Research and Development Fort Monmouth, New Jersey			10
Fort Monmodul, New Jersey		Barnes Engineering Company 30 Commerce Road	
Commanding General	64	Stamford, Connecticut	
U.S. Army Materiel Command		ATTN: H. Yates	79
Washington 25, D. C.		111111111111111111111111111111111111111	
Common de la		Battelle Memorial Institute	
Commander		505 King Avenue	
U. S. Army Missile Command Redstone Arsenal, Alabama		Columbus 1, Ohio	80
ATTN: AMSMI-RB	65	ATTN: Battelle-DEFENDER	00
ATTN: AMSMI-RRX	66	Bell Telephone Laboratories, Inc.	
ATTN: AMSMI-RNR	67	Whippany, New Jersey	
ATTN: AMCPM-ZER-R	68	ATTN: C.W. Hoover, Room 2B-105	81
0		ATTN: C. E. Paul	82
Security Office, Army Missile Command Pacific Field Office		ATTN: John McCarthy	83
Box 56, Navy 824		Dondin Composition	
c/o FPO, San Francisco, California		Bendix Corporation Systems Division	
ATTN: Dr. S. Edelberg	69	3300 Plymouth Road	
		Ann Arbor, Michigan	
Commanding General		ATTN: Systems Analysis and Math Dept	. 84
U. S. Army Research and Development		ATTN: Flight Sciences Department	85
Washington 25, D. C. ATTN: Intl. Division	70	Desire At a C	
ATTN: Physical Sciences Division	70 71	Boeing Airplane Company P. O. Box 3707	
	••	Seattle, Washington	
Commanding Officer		ATTN: Org. 2-5732/J. Klaimon	86
U.S. Army Signal Missile Support Agency			
White Sands Missile Range, New Mexico	70	Brown Engineering Company	
ATTN: SIGWS-MM-1 ATTN: MEW	72 73	Huntsville, Alabama	
ATTM: MEW	13	ATTN: Technical Library	87
U.S. Army Technical Intelligence Agency		California Institute of Technology	
Arlington Hall Station		Pasadena, California	
Arlington 12, Virginia		ATTN: Prof. L. Lees	88
ATTN: ORDLI	74		
ARO, Inc.		Central Intelligency Agency	
von Karman Facility		2930 E Street, N. W. Washington, D. C.	
Tullahoma, Tennessee		ATTN: OCR Standard Distribution	89-91
ATTN: J. Lukasiewicz	75	and a profit tourion	00 01

DISTRIBUTION LIST (continued)

Recipient	Copy No.	Recipient	Copy No
Communication and Propagation Laborate	ory	Geophysics Corporation of America	123
Stanford Research Institute		Burlington Road	
Menlo Park, California		Bedford, Massachusetts	
ATTN: Mr. Ray L. Leadabrand, Head			
Propagation Group	92	Heliodyne Corporation	124
ATTN: Dr. Carson Flammer	93	2365 Westwood Blvd	
B 4 1 5 5		Los Angeles 64, California	
Defense Documentation Center	94 - 113		
Cameron Station		Institute for Defense Analyses	
Alexandria, Virginia		1666 Connecticut Avenue N. W.	
		Washington 9, D.C.	
Cornell Aeronautical Laboratory		ATTN: Dr. J. Menkes	125
4455 Genesee Street		ATTN: Dr. L. Biberman	126
Buffalo 21, New York		ATTN: Dr. R. Fox	127
ATTN: J. Lotsof	114	ATTN: Dr. J. Martin	128
ATTN: W. Wurster	115	ATTN: Mr. D. Katcher, JASON Library	129
ATTN: Applied Physics Dept.	116		
D. C. D. L. C. L.		Institute of Science and Technology	
Defense Research Corporation		The University of Michigan	
6300 Hollister Avenue,		P. O. Box 618	
Goleta, California		Ann Arbor, Michigan	
ATTN: W. Short	117	ATTN: BAMIRAC Library	130
Director		Jet Propulsion Laboratory	
Electromagnetic Warfare Laboratory		4800 Oak Grove Drive	
Wright-Patterson Air Force Base		Pasadena, California	
Dayton, Ohio		ATTN: H. Denslow	131
ATTN: ASRN/W. Bahret	118	ATTN: Library	132
Electro-Optical Systems, Inc.		Kaman Nuclear Division	
300 N. Halstead Street		Colorado Springs, Colorado	
Pasadena, California		ATTN: A. Bridges	133
ATTN: R. Denison	119	Drages	100
		Director	
General Applied Sciences Laboratories		Langley Research Center	
Merrick and Stewart Avenues		Langley Field, Virginia	
Westbury, Long Island, New York		ATTN: W. Erickson	134
ATTN: M. Bloom	120	ATTN: R. L. Trimpi	135
General Dynamics Corporation		Lockheed Corporation	
Astronautics Division		Missiles and Space Division	
San Diego, California		Sunnyvale, California	
ATTN: Chief Librarian, Mail Zone 6-157	121	ATTN: Ray Munson	136
General Electric Company		Molnon Inc	
Re-entry Vehicles Division		Melpar, Inc.	
3198 Chestnut Street		Applied Science Division	
Philadelphia, Pennsylvania		11 Galen Street Watertown 72, Massachusetts	
ATTN: L. I. Chaseen, Room 3446	122	ATTN: Librarian	137
		TO A COLUMN A CLASSICIA	

DISTRIBUTION LIST (continued)

Recipient	Copy No.	Recipient	Copy No
Martin Aircraft Company		Purdue University	
Orlando, Florida		School Aero and Engineering Sciences	
ATTN: J. Mays	138	La Fayette, Indiana	
		ATTN: I. Kvakovsky	151
Director		ATTM. I. KVAKOVSKY	151
Marshall Space Flight Center		Padio Componetion of America	
Huntsville, Alabama		Radio Corporation of America	152
ATTN: M-AERO-TS	1.00	Missiles and Surface Radar Division	
MIIN. M-ALNO-15	139	Moorestown, New Jersey	
Massachusetts Institute of Technology		Th- 0- 10- 11	
Lincoln Laboratory		The Rand Corporation	
P. O. Box 73		1700 Main Street	
		Santa Monica, California	
Lexington 73, Massachusetts		ATTN: Library	153
ATTN: M. Herlin	140		
ATTN: R. Slattery	141	Raytheon Manufacturing Company	
ATTN: V. Guethlen	142	Missile Systems Division	
		Bedford, Massachusetts	
Chief		ATTN: I. Britton, Librarian	154
U.S. Navy Bureau of Weapons		i. Ditton, Dividian	134
Washington 25, D. C.		Roma Air Davalones and Continu	
ATTN: RMWC-322	140	Rome Air Development Center	
1111. 1131111 0 022	143	Griffiss Air Force Base	
Chief of Naval One water		Rome, New York	
Chief of Naval Operations		ATTN: P. Sandler	155
Washington 25, D. C			
ATTN: OP-07T10	144	The Martin Company	
		Aerospace Division, Mail No. T-38	7
Commander		P.O. Box 179, Denver, Colorado 8020	1
U.S. Naval Ordnance Laboratory		ATTN: R. E. Compton, Jr.	156
White Oak, Silver Spring, Maryland			150
ATTN: Technical Library	145	Space Technology Laboratories, Inc.	
,		1 Space Park	
Director		Redondo Beach, California	
U. S. Naval Research Laboratory		ATTN: Leglie Harman	150
Washington 25, D. C.		ATTN: Leslie Hromas	157
ATTN: Code 2027	1.40	m	
ATTN. Code 2021	146	The Warner and Swasey Company	158
AT TT 1 TT 1		Control Instrument Division	
New York University		32-16 Downing Street	
Department of Aero Engineering		Flushing 54, New York	
University Heights			
New York 53, New York		University of California	
ATTN: L. Arnold	147	San Diego, California	
		ATTN: Prof. N. M. Kroll	159
North American Aviation			100
Space and Information Systems Division		University of California	
12214 Lakewood Blvd		Lawrence Radiation Laboratory	
Downey, California		Livermore, California	
ATTN: E. Allen	148	ATTN: C. Craig	160
	140	min. c. craig	160
Princeton University		Scientific and Technical	
Princeton, New Jersey		Information Facility	
ATTN: Prof. E. Frieman	140	P. O. Box 5700	
ATTN: Prof. S. Bogdonoff	149	Bethesda, Maryland	
ATIM. FIOI. B. Dogaonon	150		61, 162

DISTRIBUTION LIST (Concluded)

Recipient	Copy No.	Recipient	Copy No.
General Electric Company Re-entry Systems Department Missile and Space Division P. O. Box 8555 Philadelphia		* Commanding Officer U. S. Army Missile Command Redstone Arsenal, Alabama ATTN: SMIDW-B1 (C)	168
Philadelphia, Pennsylvania ATTN: Mr. H. W. Ridyard	163	**British Joint Mission British Embassy	
University of Michigan Radiation Laboratory 201 Catherine Ann Arbor, Michigan		3100 Massachusetts Avenue, N. W. Washington, D. C. ATTN: Mr. F. I. Reynolds, Defense Research Staff	169
ATTN: R. J. Leite Valley Forge Space Technical Center General Electric Company P. O. Box 8555 Philadelphia 1, Pennsylvania	164	**Australian Embassy 2001 Connecticut Avenue N. W. Washington, D. C. ATTN: D. Barnsley, Defense Research and Development Rep.	170
ATTN: J. Farber	165	*University of Toronto Department of Electrical Engineer	171
Director Weapons Systems Evaluation Group Pentagon, Room 1E-800	166	Toronio, Ostario, Canada ATTN: Mr. H. Treial, Research	
Washington 25, D. C. U. S. Army Liaison Office Canadian Armament Research and Development Establishment P. O. Box 1427		Capt. L. L. Schoen, USAF USAF Technical Representative c/o GM Defense Research Laboratories	172
Quebec, P. Q., Canada ATTN: Lt. Col. E. W. Kreischer	167	GM Defense Research Laboratories	172 and
ATTN. Et. Col. E. W. Kreischer	101		above

- * Unclassified reports only.
- ** Research reports on hypervelocity ranges and air and contaminant chemistry only not included for analytical reports on U.S. missile data or classified reports.

Additional Distribution for Semiannual Reports only:

Office of Naval Research
Department of the Navy
Washington 25, D. C.
ATTN: Dr. S. Silverman, Science Director 1 copy
ATTN: Dr. F. Isakson, Physics Branch 1 copy
ATTN: Mr. M. Cooper, Fluid Dynamics Branch 1 copy

UNCLASSIFIED

UNCLASSIFIED